



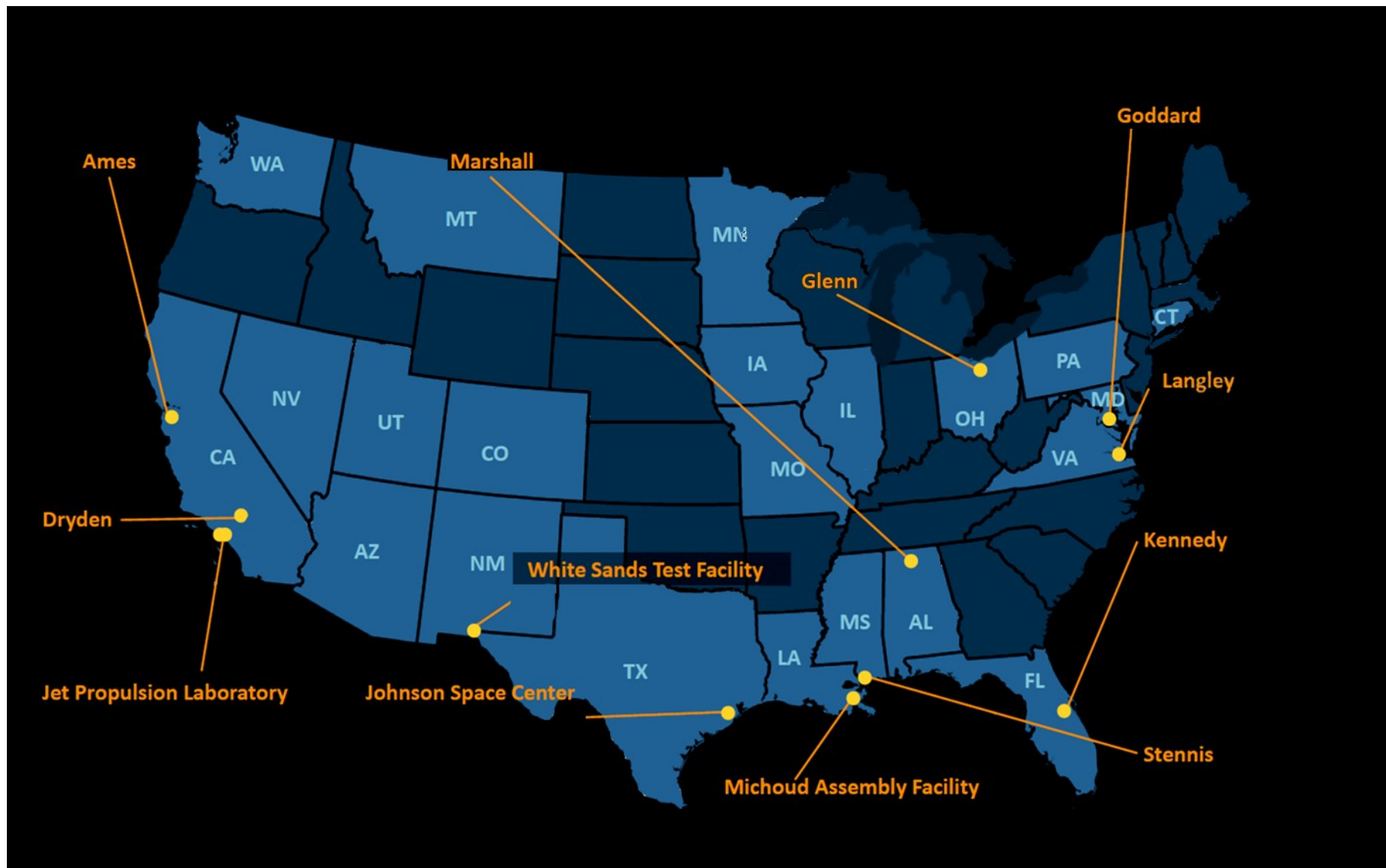
# **Overview of the Space Environment Test Facilities at NASA GRC's Plum Brook Station**

**2019 IEEE WiSEE/PWST**  
October 16<sup>th</sup>, 2019

**Richard K. Evans**  
**(NASA GRC)**

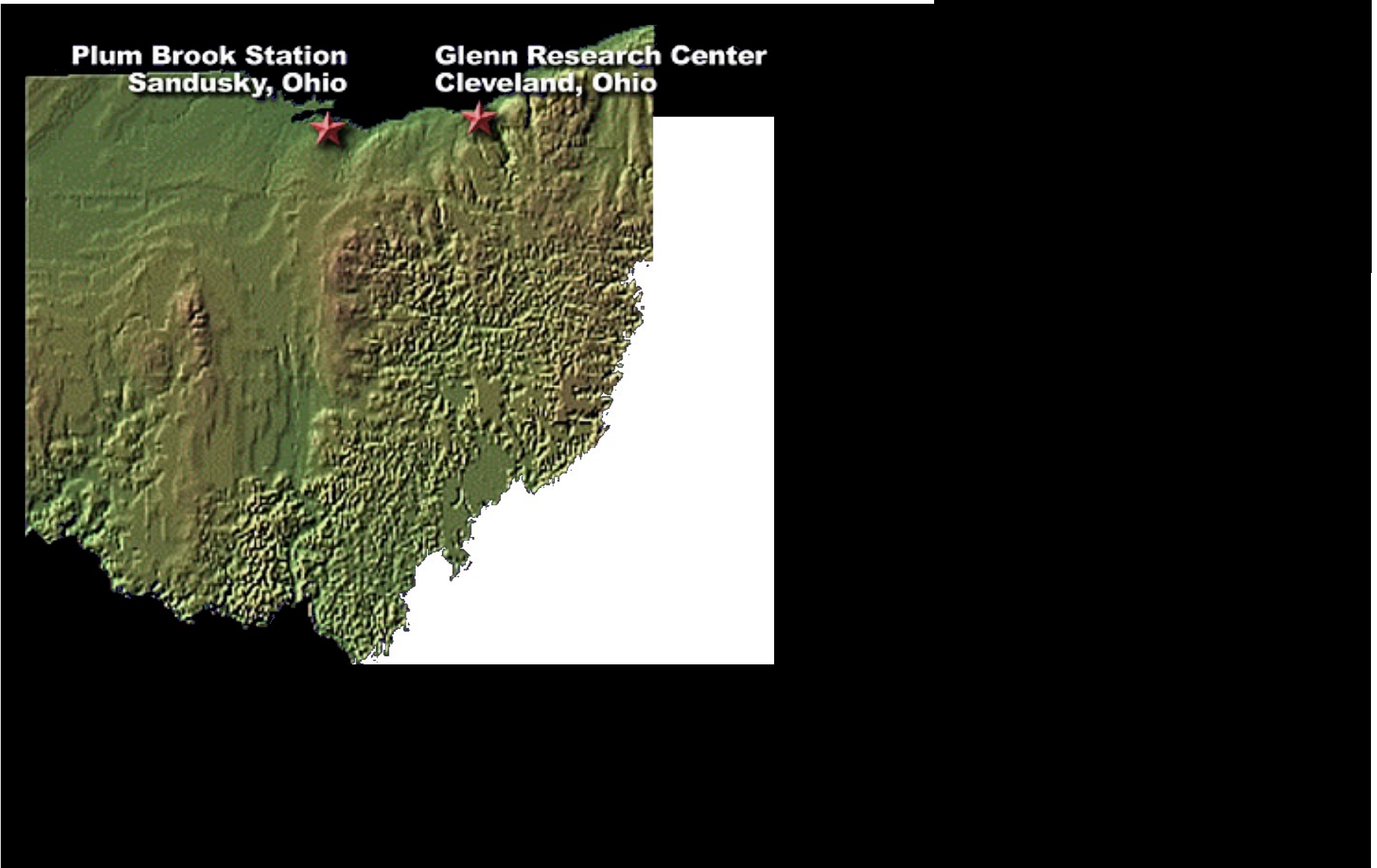


# NASA Field Centers





# NASA in Ohio







# NASA's Plum Brook Station in Sandusky, Ohio



**Satellite campus of NASA GRC used for large-scale space environment testing**





# NASA's Plum Brook Station in Sandusky, Ohio

## What makes Plum Brook Station Unique?

- 10 square miles (25.9 km<sup>2</sup>) / 6400 acres / 5400 acres fenced
  - Secure site with securable data and facilities
- Staffed and equipped for
  - Large scale cryogenic tests in harsh conditions
  - Extreme temperatures, pressures, noises and vibrations
  - Explosives and hazardous materials storage
- 100 megawatt power grid
  - Affordable power (vs commercial rates) on day shift
  - High MW tests on site don't affect local users
- Large scale availability of gas and water
- 41/50 miles (66/80 km) from large airports (MFD/CLE)
- 7 miles (11.2 km) from deep water Lake Erie port of Huron, OH
- Multiple unique, world-class, space environment test facilities



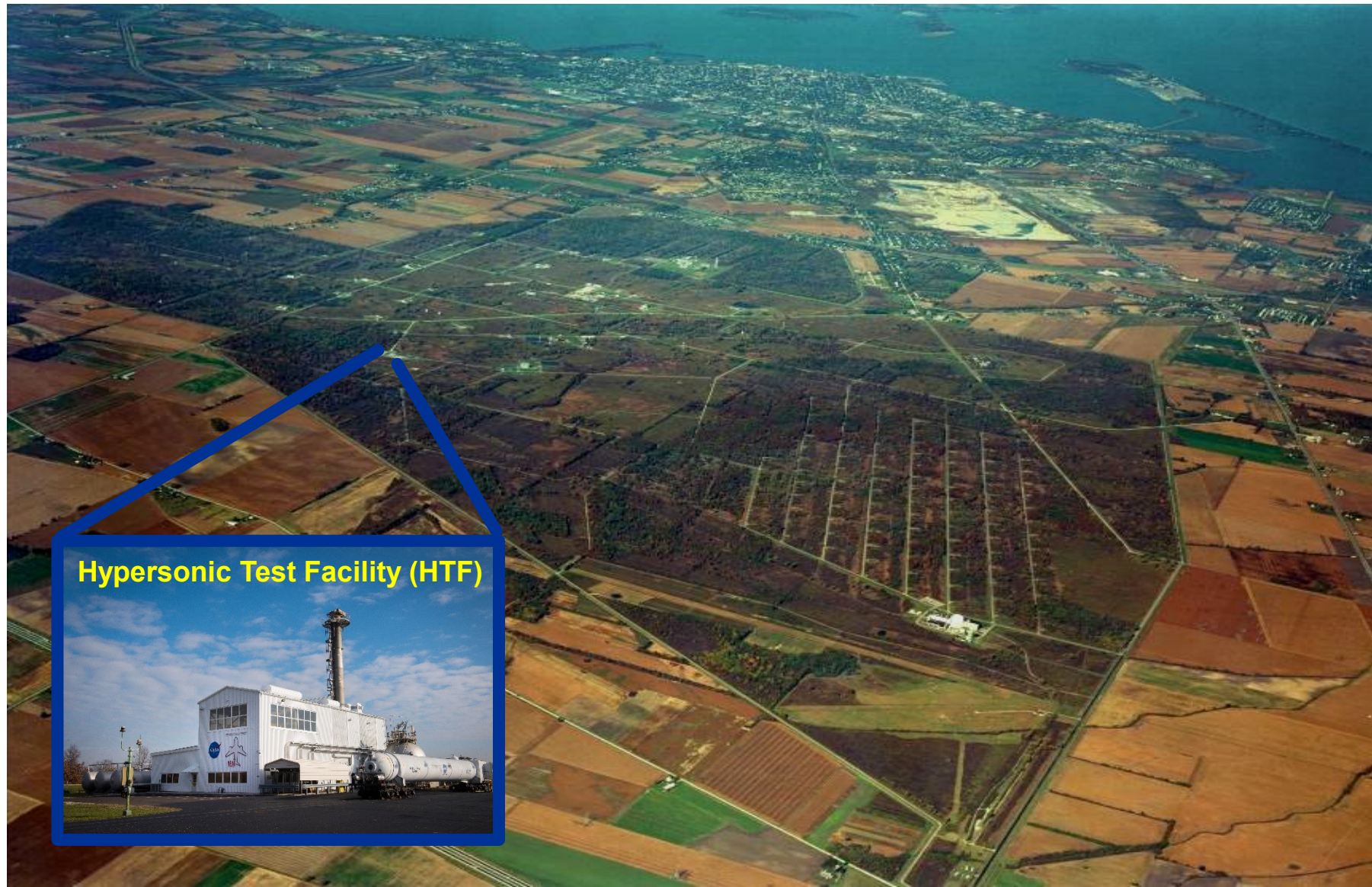
# NASA's Plum Brook Station in Sandusky, Ohio

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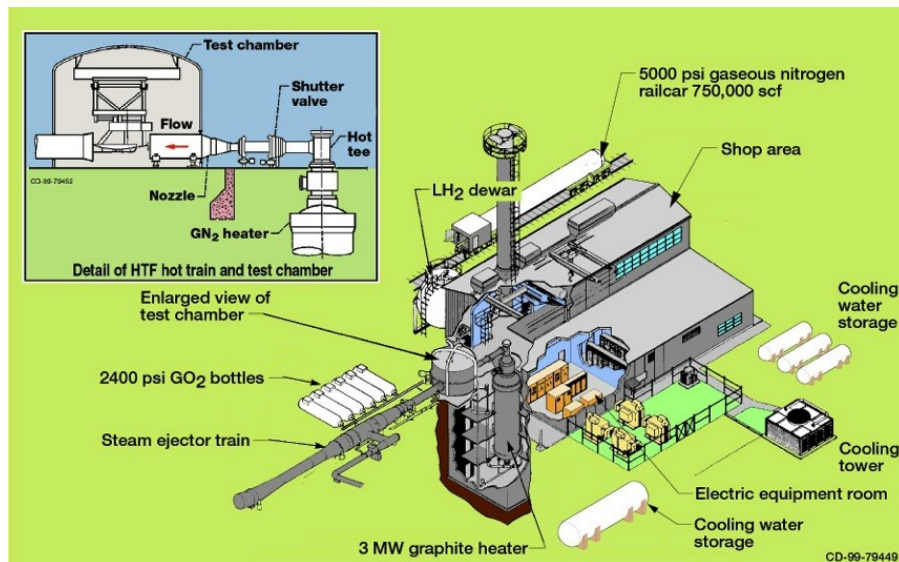
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*“Where Space Comes Down to Earth”*

# NASA's Plum Brook Station in Sandusky, Ohio







- **HTF is the United States' only large scale, non-vitiated (clean air) hypersonic test facility**
  - 42” test section
  - Run times up to 5 minutes
  - Capable of **Mach 5, 6, 7 flows** in either free-jet or direct connect modes
- **This facility provides basic hot-flow test capabilities**
  - Large scale hot-gas rocket nozzle testing (ambient or altitude up to **120k ft / 36km** )
  - Materials testing
  - **Communications through ionized flows for Entry, Descent, Landing**
  - Hybrid gas electric propulsion for aviation
  - **Hypersonic sensors**, propulsion, airframe/engine integration

<https://www1.grc.nasa.gov/facilities/hf/>



# NASA's Plum Brook Station in Sandusky, Ohio



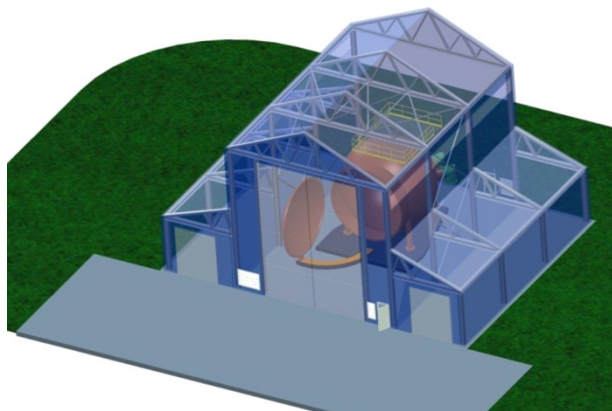
**Combined Effects Chamber (CEC)**



# Combined Effects Chamber (CEC)

## Planet/Planetoid Space Simulation

CEC helps rule out problems in space surface environments (solid as well as liquid) and qualify cryogenically cooled sensors for aeronautic use



### KEY FEATURES

- Large size (25' dia. Spherical chamber)
- Pressure (**35 psi**) and
- vacuum capability (**10-8 torr**)
- Liquid hydrogen cold wall (**40K temps**)
- It can accommodate regolith simulant in the chamber, including stratification for drilling & excavation demonstrations
- Built-in shaker system that can be used for settling and stratification
- Simple design makes chamber inexpensive to maintain and operate

### SYSTEMS AND COMPONENT TESTING FOR:

- Landers (robotic and human)
- Rovers (robotic and human)
- In-situ Resource Utilization (ISRU) systems
- Surface power systems
- Heat radiators and deployment systems
- Surface habitats
- EVA systems
- And much more





# NASA's Plum Brook Station in Sandusky, Ohio



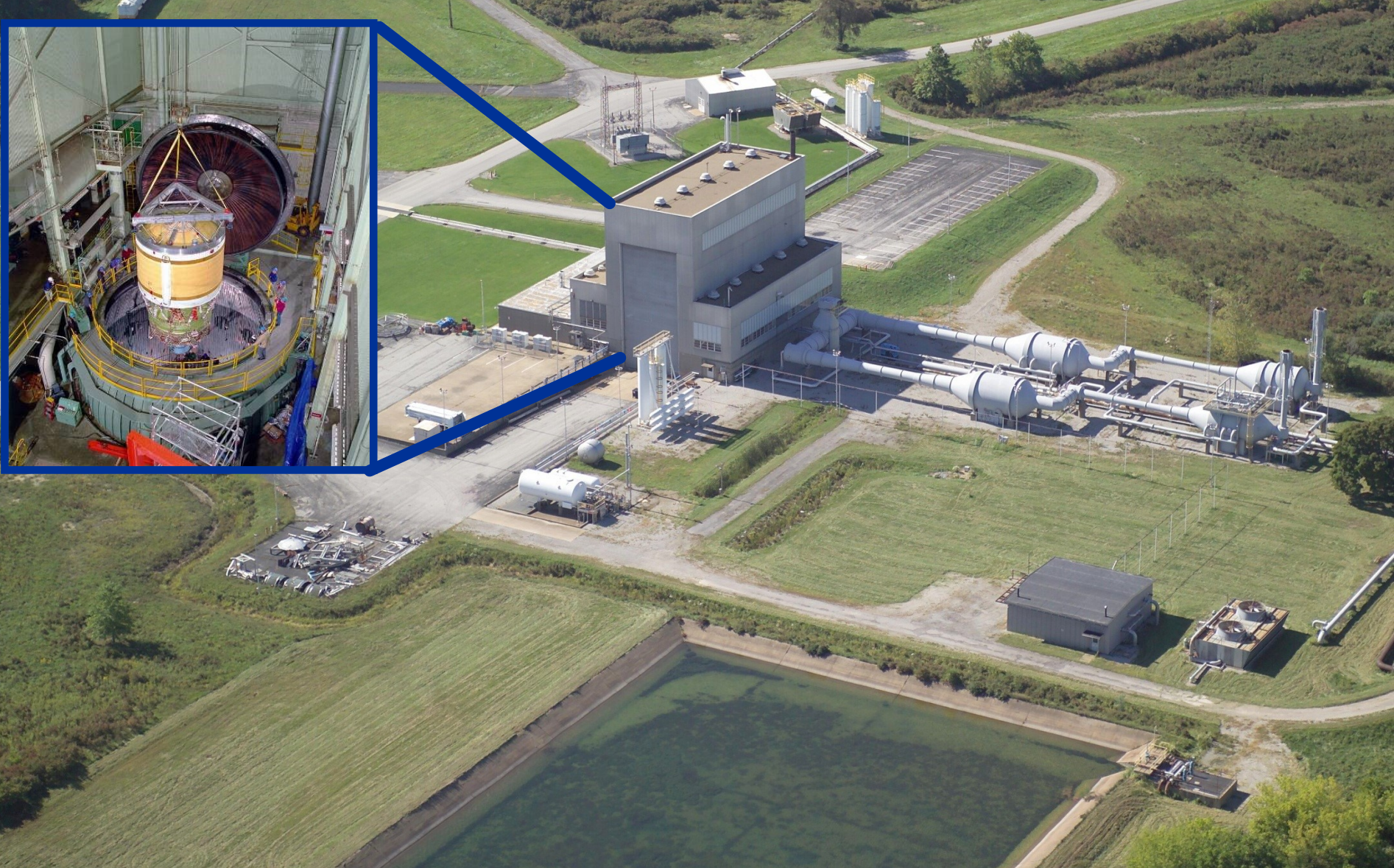


# NASA Glenn Research Center In-Space Propulsion (ISP) Facility



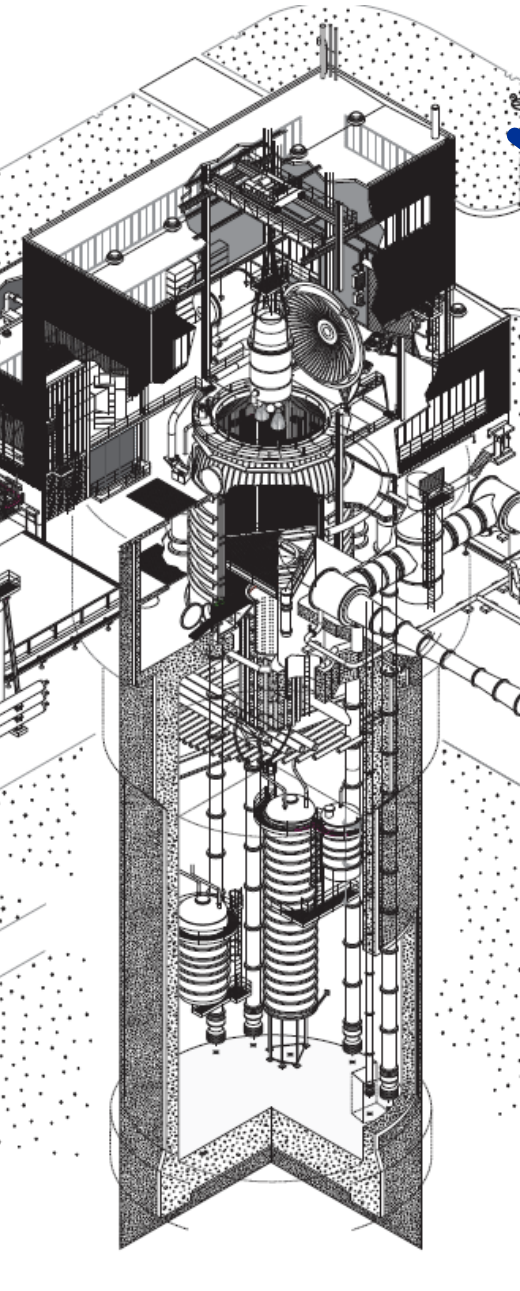


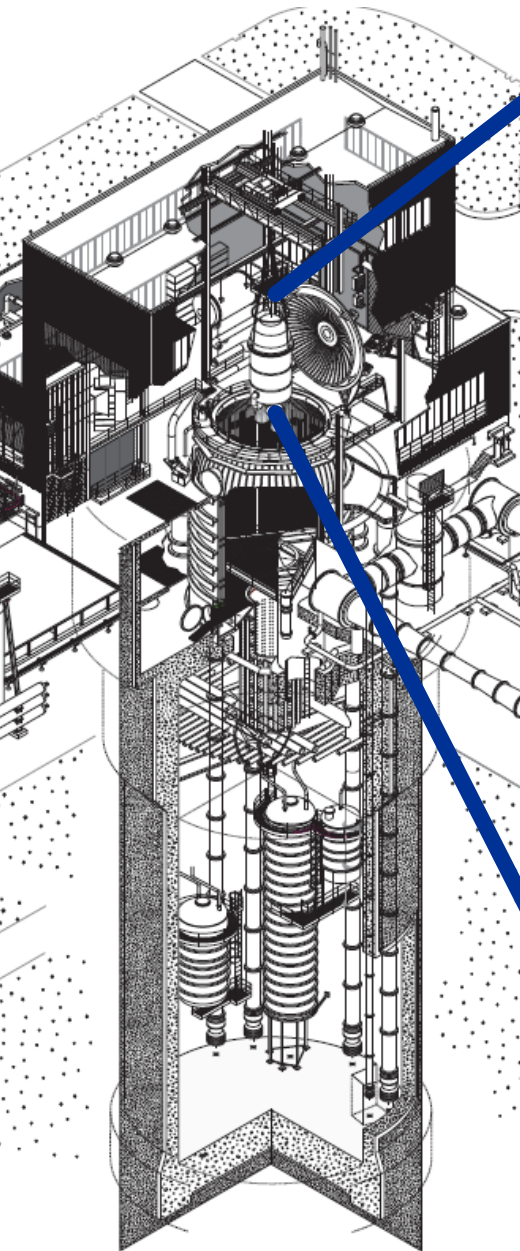
# NASA Glenn Research Center In-Space Propulsion (ISP) Facility





# NASA Glenn Research Center In-Space Propulsion (ISP) Facility





SpaceX  
@SpaceX

Follow



Crew Dragon is at @NASA's Plum Brook Station testing facility in Ohio, home to the largest thermal vacuum chamber in the world, to demonstrate its capability to withstand the extreme temperatures and vacuum of space. [instagram.com/p/BkQ8w0mF0xa](https://www.instagram.com/p/BkQ8w0mF0xa)



4:33 PM - 20 Jun 2018

2,668 Retweets 16,933 Likes



312 2.7K 17K



# ISP Facility Capability Summary

(all values are approximate and for reference only)



## Rocket Test Capability with Vacuum Start

(assumes LH2/LO2 props\*)

10K lb Thrust Engine for 380 sec Test Duration

100K lb Thrust Engine for 270 sec Test Duration

300K lb Thrust Engine for 20 sec Test Duration

500K lb Thrust Engine for 5 sec Test Duration

\* Other propellant combinations are possible

## Space Simulation

Test chamber is capable of vacuum space soak at  $5 \times 10^{-8}$  torr with -320°F (-195 °C) cold wall and 130w/ft<sup>2</sup> radiant heating. Uses mechanical and oil diffusion pumps to achieve vacuum conditions. Facility LN<sub>2</sub> capacity is 200gal/min (cold wall uses 100gal/min).

## Physical Dimensions

Test Chamber Clear Space    33ft (10m) dia  
55ft (16.7m) height

Top Entrance                    27ft (8.2m) dia

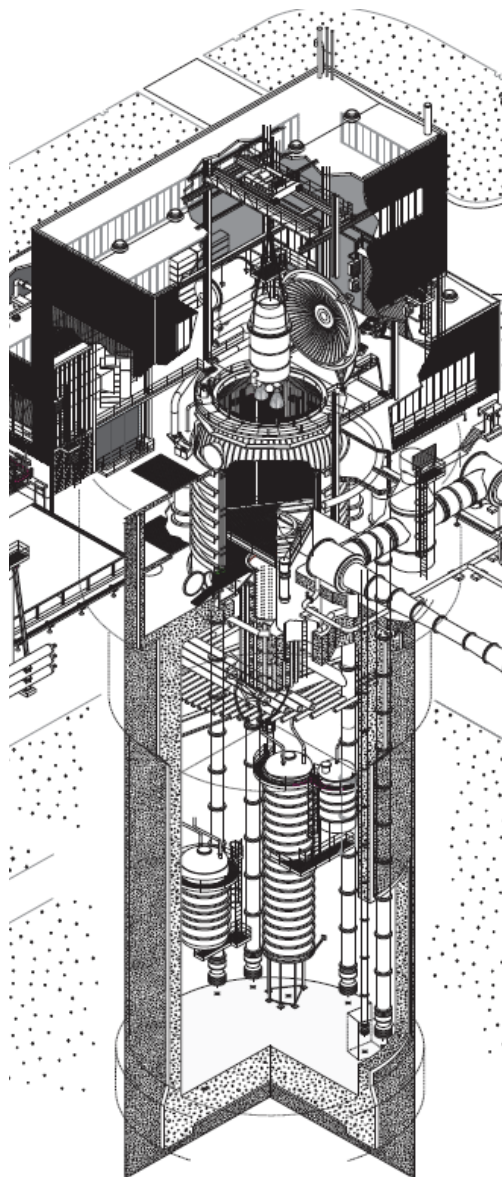
Diffuser                        11ft dia x 37ft long

Spray Chamber                67ft dia x 120ft (36.5m) deep

## Exhaust System

Maximum capacity of 120 lbs/sec Dry Air Equivalent (DAE) at 1psia using a three stage intercondensing steam ejector consuming 348 lb/sec of steam.

Produces ~0.14 psia in spray chamber with 40°F water.





# NASA's Plum Brook Station in Sandusky, Ohio

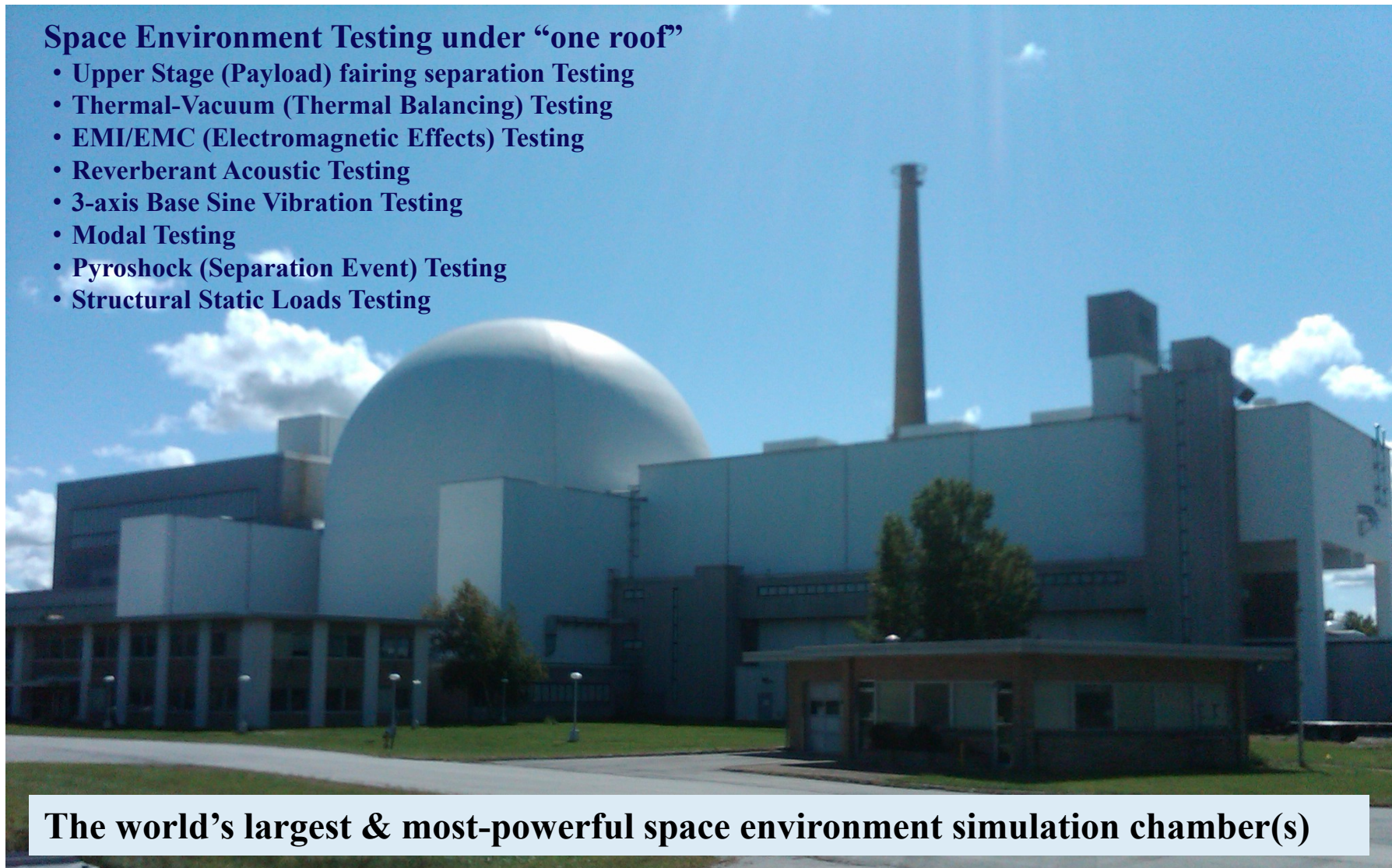




# Space Environments Complex (SEC)

## Space Environment Testing under “one roof”

- Upper Stage (Payload) fairing separation Testing
- Thermal-Vacuum (Thermal Balancing) Testing
- EMI/EMC (Electromagnetic Effects) Testing
- Reverberant Acoustic Testing
- 3-axis Base Sine Vibration Testing
- Modal Testing
- Pyroshock (Separation Event) Testing
- Structural Static Loads Testing



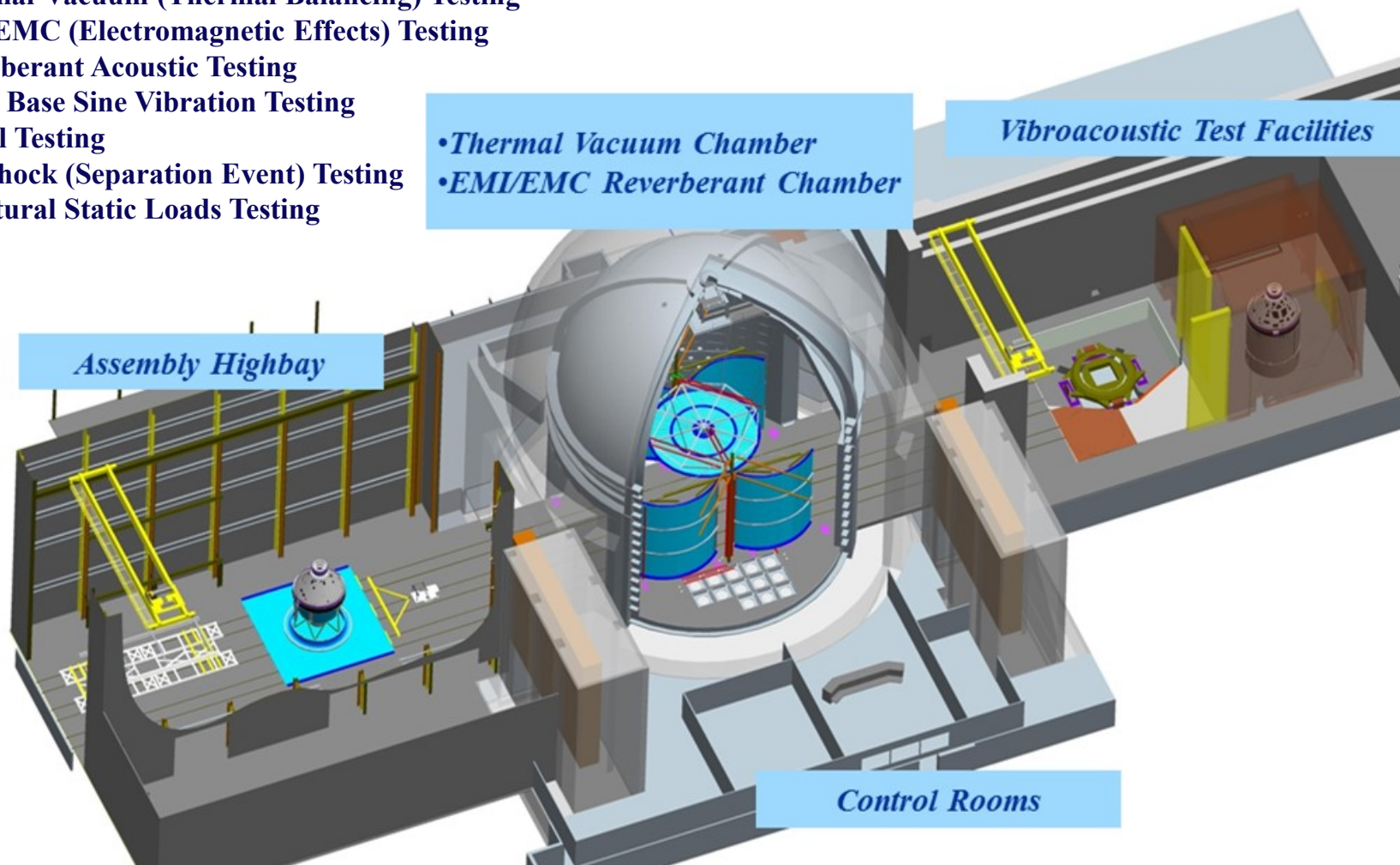
**The world's largest & most-powerful space environment simulation chamber(s)**



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# Space Environments Complex (SEC)

**One-of-a-kind environmental testing capability at ONE location:**



**Assembly/Staging  
Area (High-bay)**

**Thermal-Vacuum  
/EMI Chamber**

**Mechanical  
Vibration**

**Reverberant  
Acoustic**

<https://www.nasa.gov/specials/sec360/#>



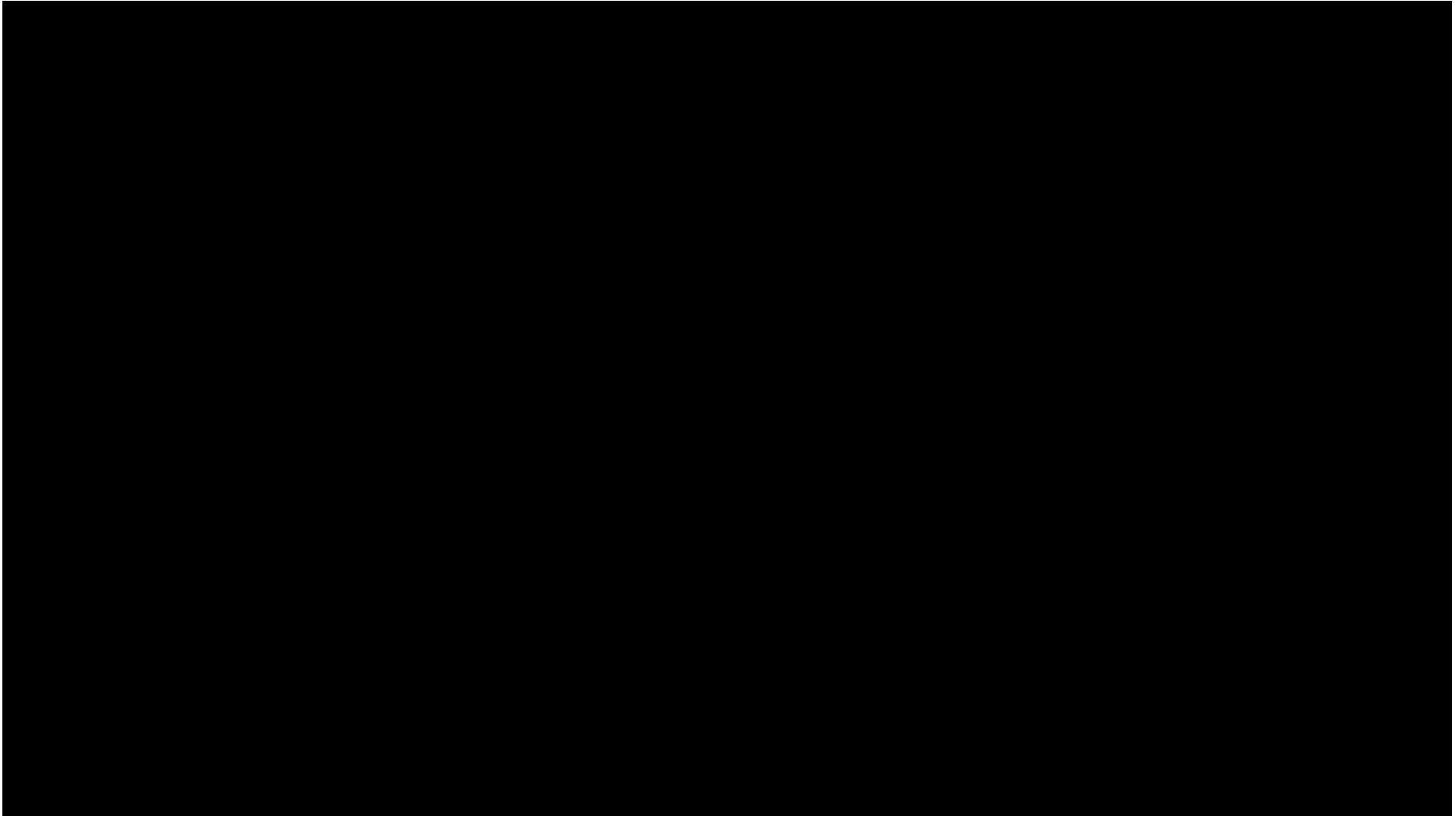
# Space Environments Complex (SEC)





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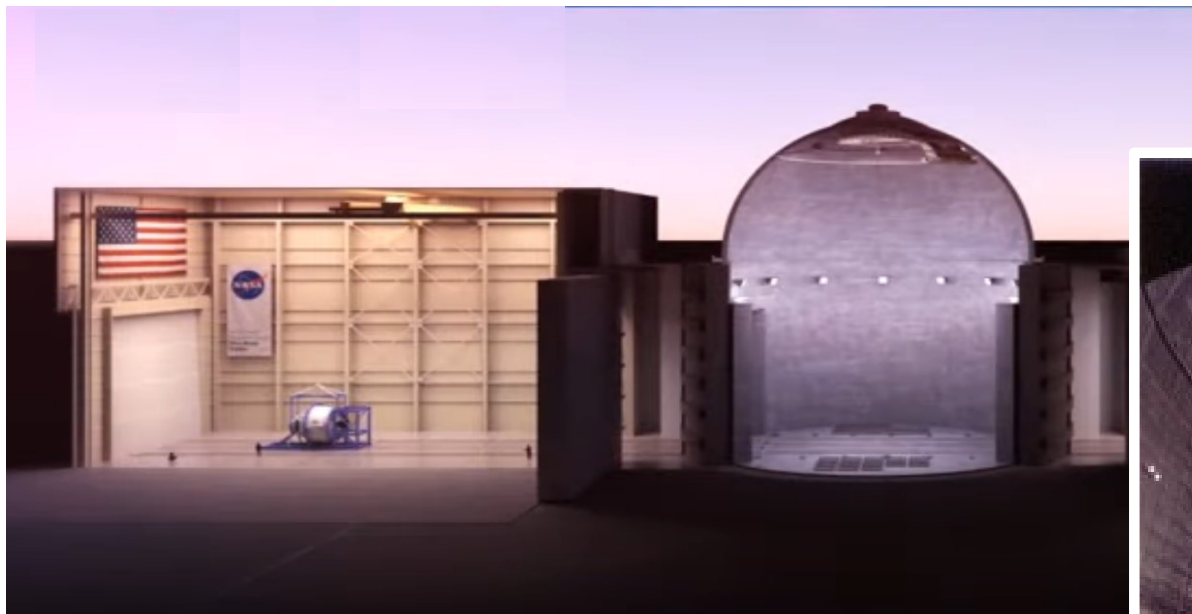
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# Space Environments Complex (SEC)

**One-of-a-kind environmental testing capability at ONE location:**



## Thermal-Vacuum / EMI Chamber

- ... the largest space simulation chamber in the world
- ✓ 800,000 ft<sup>3</sup> volume, 100ft (30m) diameter, 122ft (37m) high
- ✓ Ambient to 10<sup>-6</sup> Torr vacuum capability
- ✓ Features variable geometry cryogenic shrouds
- ✓ 7 MW power available for solar simulation
- ✓ Reverberant-mode EMI/EMC capability (45MHz–40 GHz)



# Space Environments Complex (SEC)

**One-of-a-kind environmental testing capability at ONE location:**



## Thermal-Vacuum / EMI Chamber

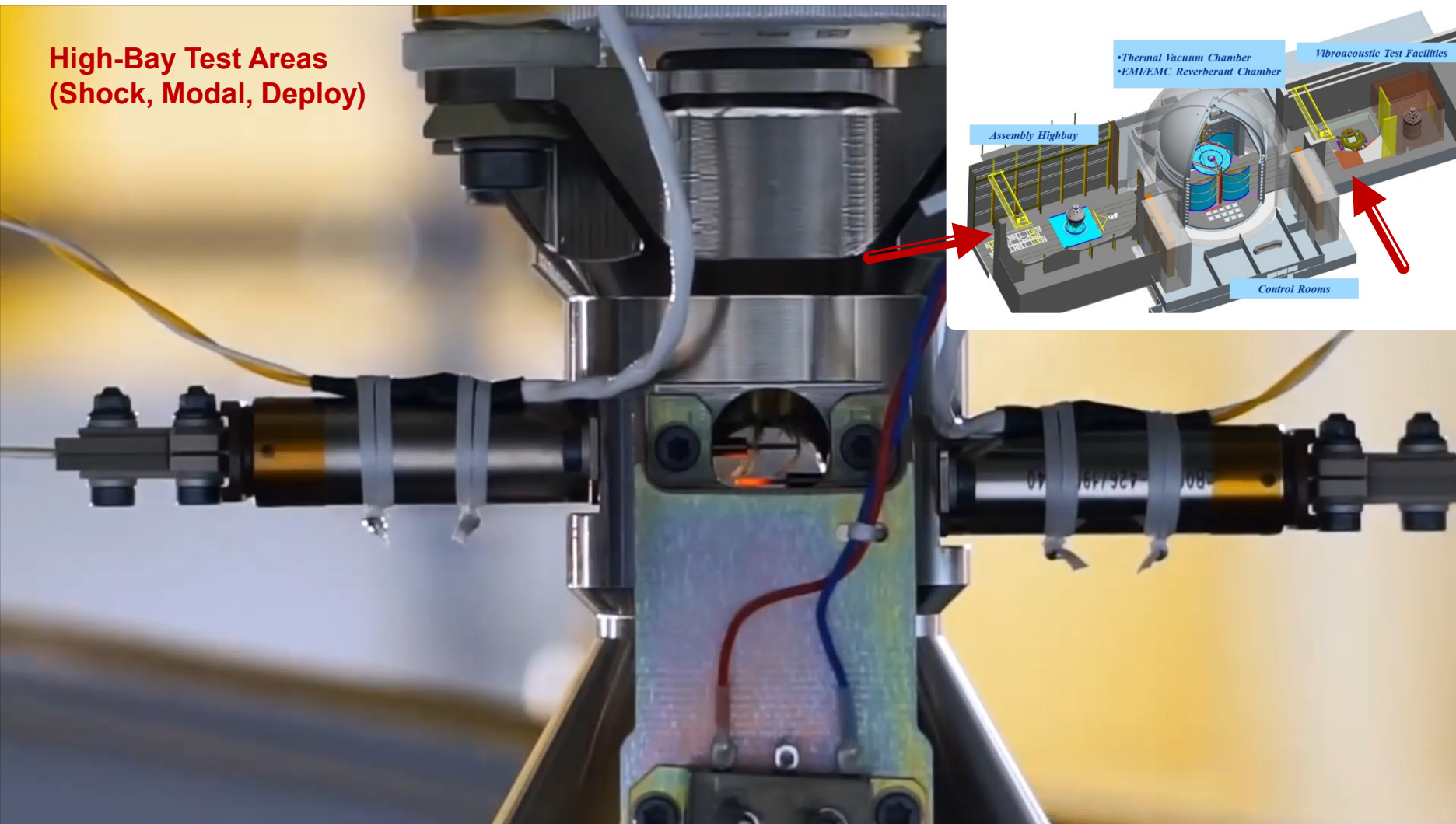
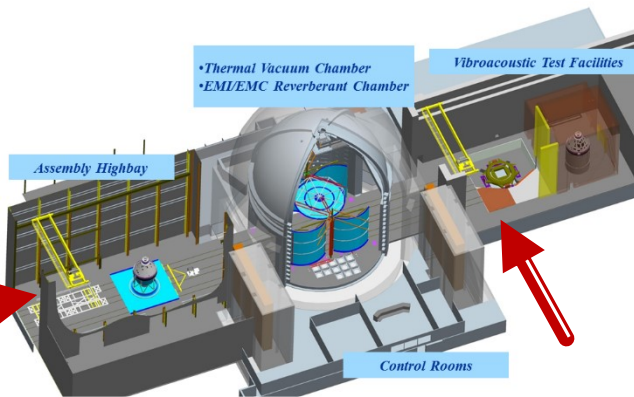
- ... the largest space simulation chamber in the world
- ✓ 800,000 ft<sup>3</sup> volume, 100ft (30m) diameter, 122ft (37m) high
- ✓ Ambient to 10<sup>-6</sup> Torr vacuum capability (-200 to 150 °C)
- ✓ Features variable geometry cryogenic shrouds
- ✓ 7 MW power available for solar simulation
- ✓ Reverberant-mode EMI/EMC capability (45MHz–40 GHz)



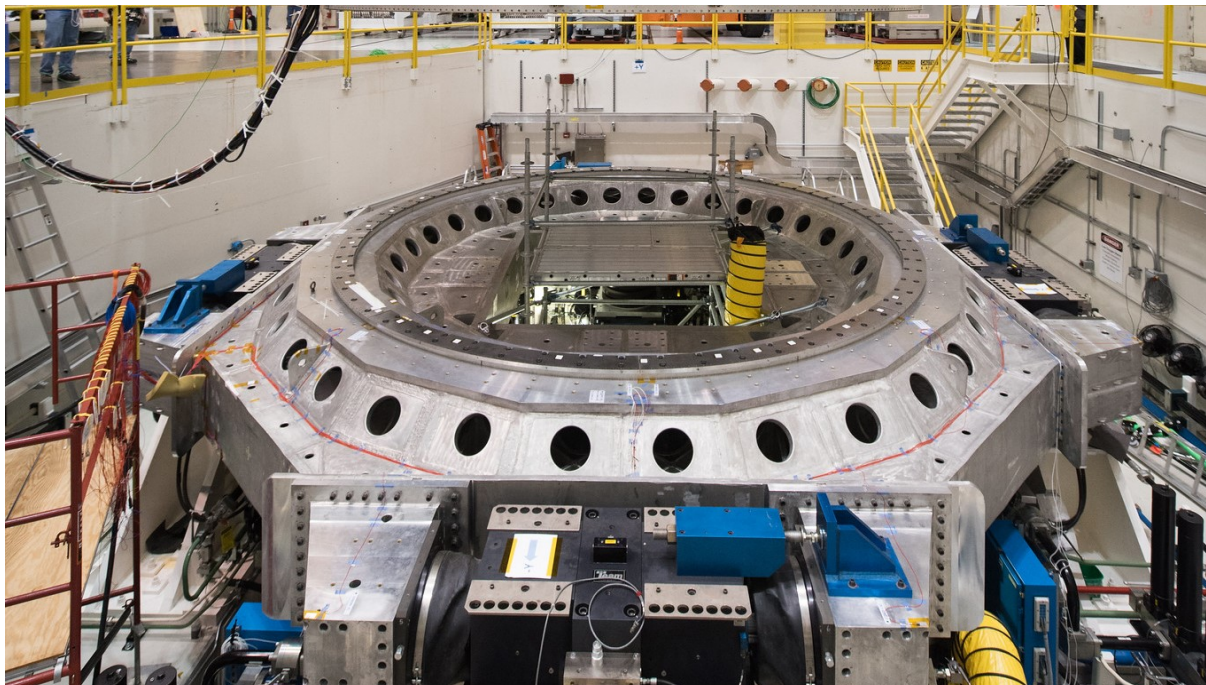
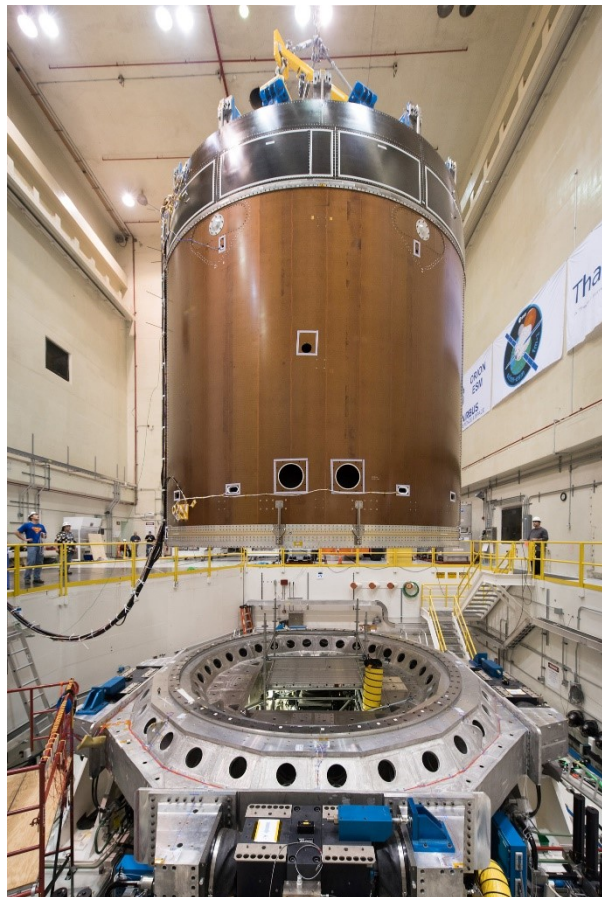


# Space Environments Complex (SEC)

**High-Bay Test Areas  
(Shock, Modal, Deploy)**



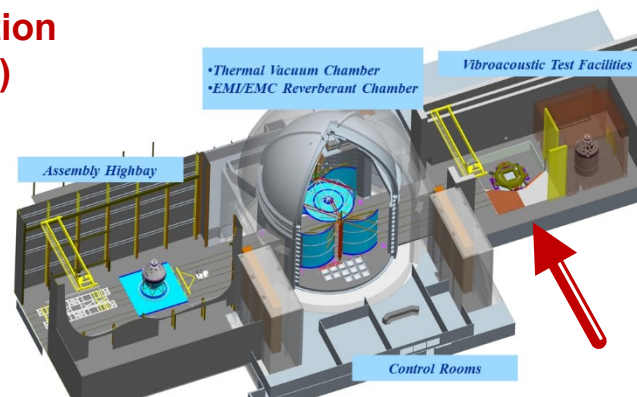
# Space Environments Complex (SEC)



## Mechanical Vibration Test Facility (MVF)

... the highest capacity shaker system in the world

- ✓ 18ft (5.4m) diameter shaker table
- ✓ 75,000 lb (34,000 kg) test article  
(1.25 g's vertical, 1.0 g's horizontal)
- ✓ Frequency range: 5 - 150 Hertz
- ✓ Sinusoidal testing in 3-axis



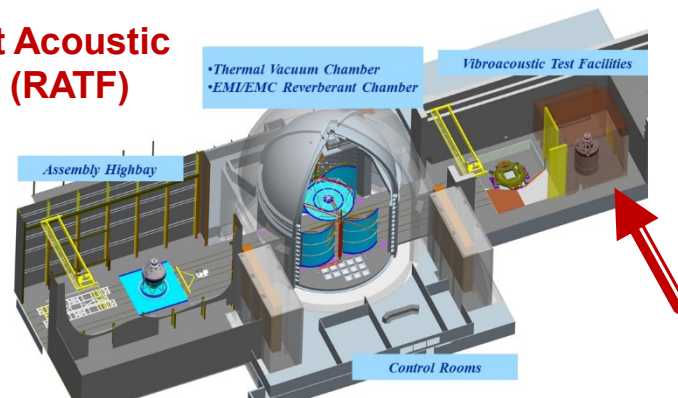


# Space Environments Complex (SEC) - RATF

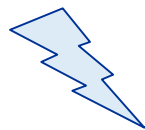


## Reverberant Acoustic Test Facility (RATF)

- ... the most powerful reverberant chamber in the world
- ✓ Capable of an overall sound pressure level of 163 dB
  - ✓ 101,000 ft<sup>3</sup> volume, 37.5 ft x 47.5 ft x 57 ft high (2,860 m<sup>3</sup> volume, 11.5m x 14.5m x 17.5m high)
  - ✓ 36 acoustic modulators and horns for highly-tailored spectrum-shaping



# “Wired” Measurement Topology



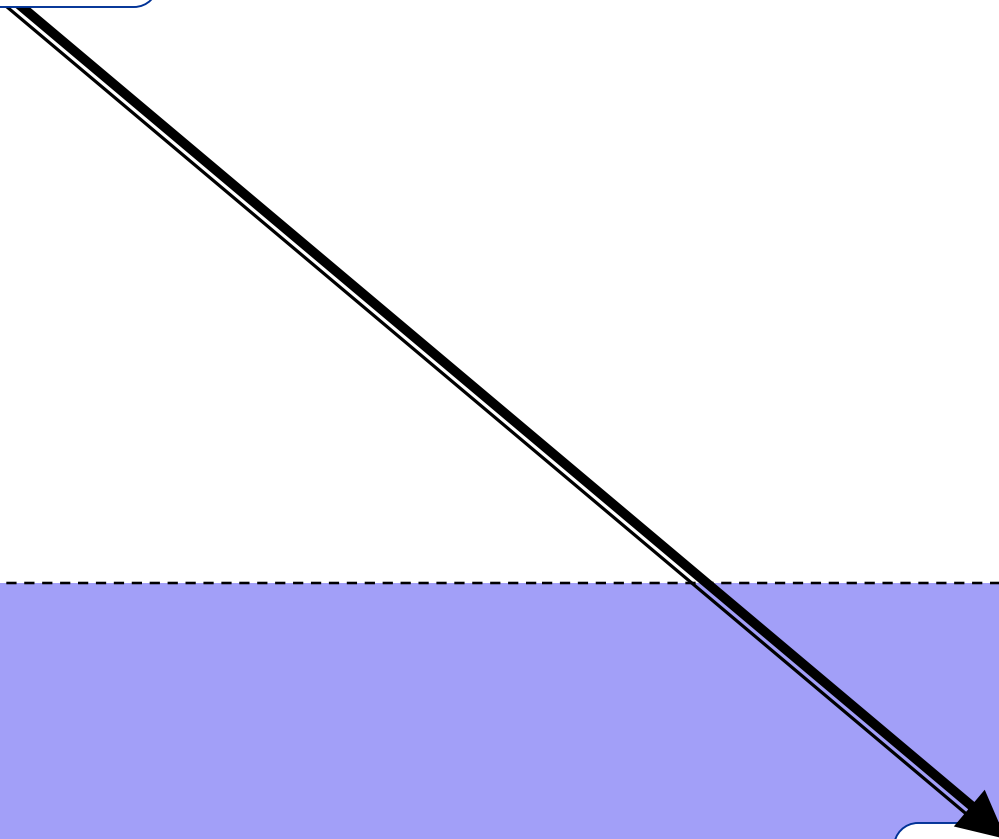
**Sensors &  
Transducers**

128 – 2,000+ channels

Accelerometers, Mics, Strain, Temperature, etc...

$\mu\text{W}/\text{nW}$

(Small Signal Regime)

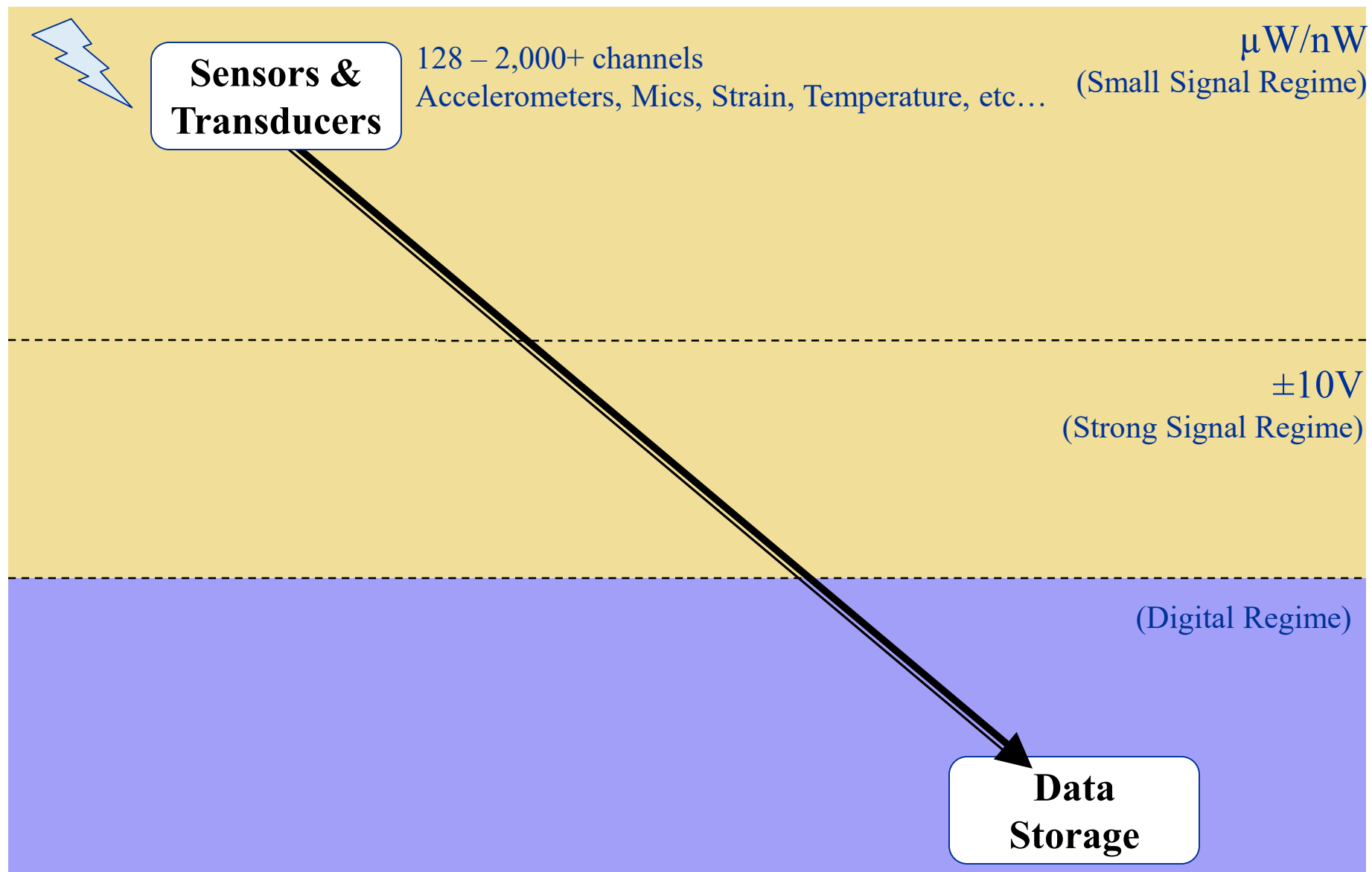


(Digital Regime)

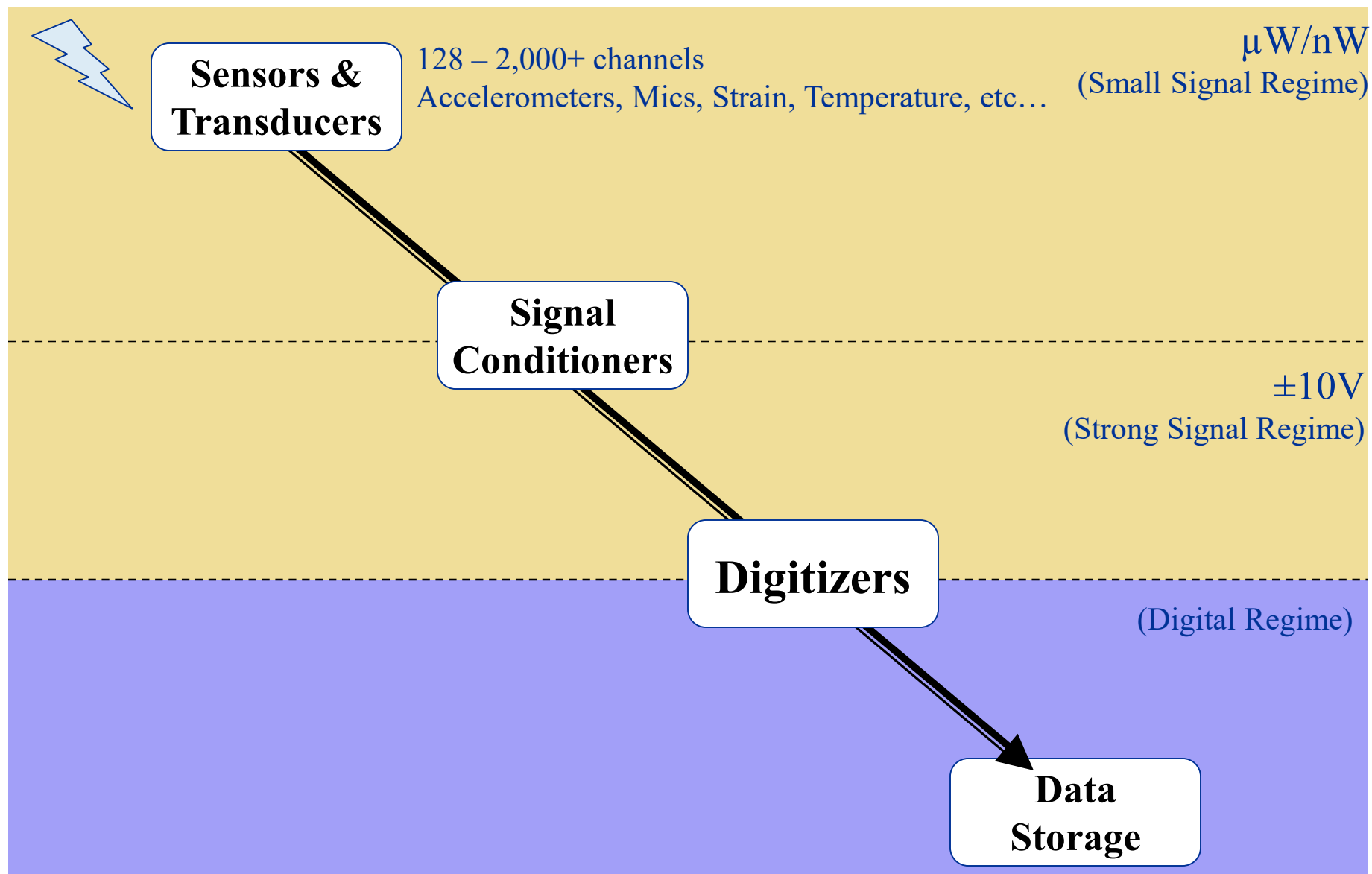
**Data  
Storage**



# “Wired” Measurement Topology

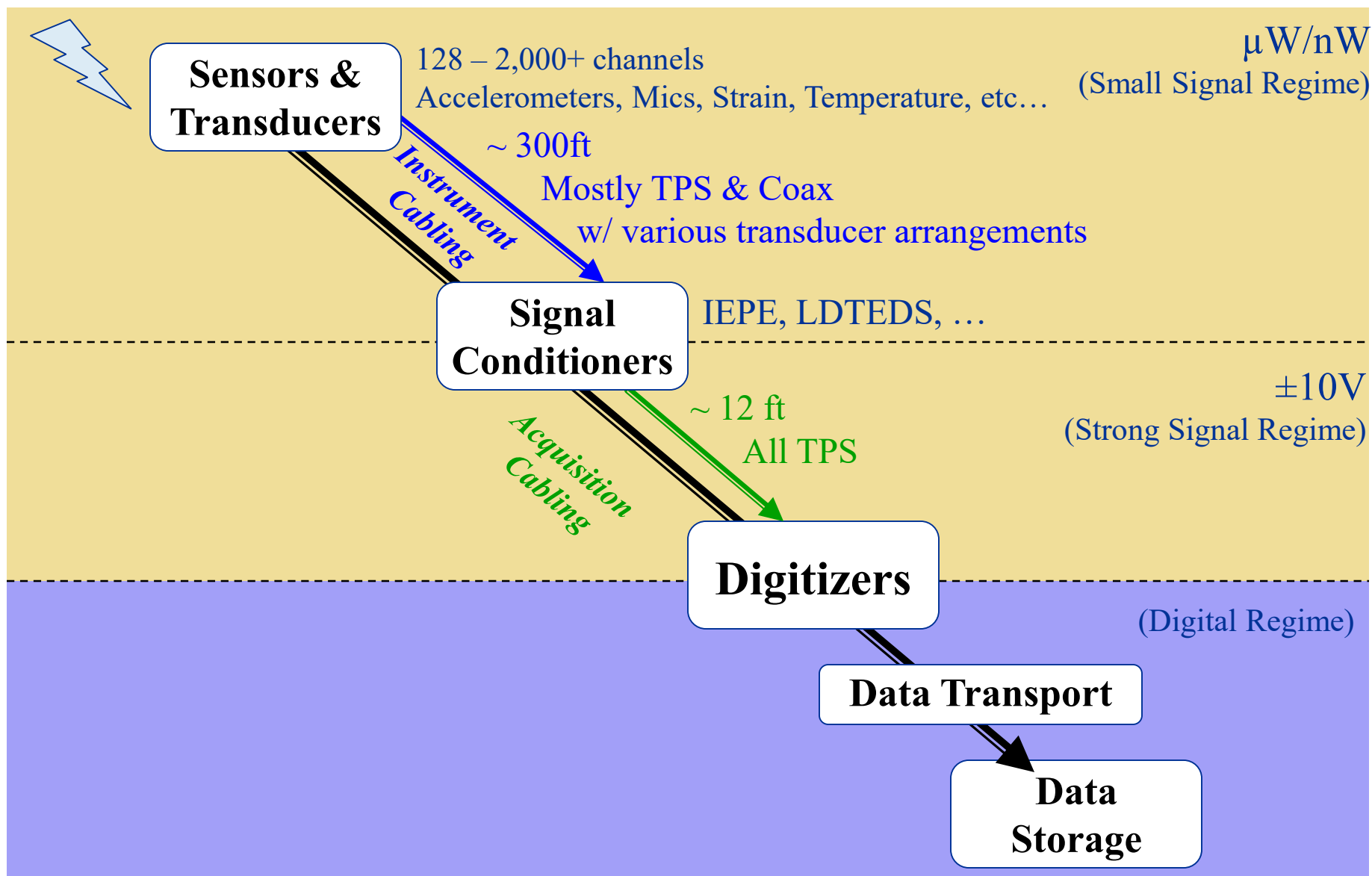


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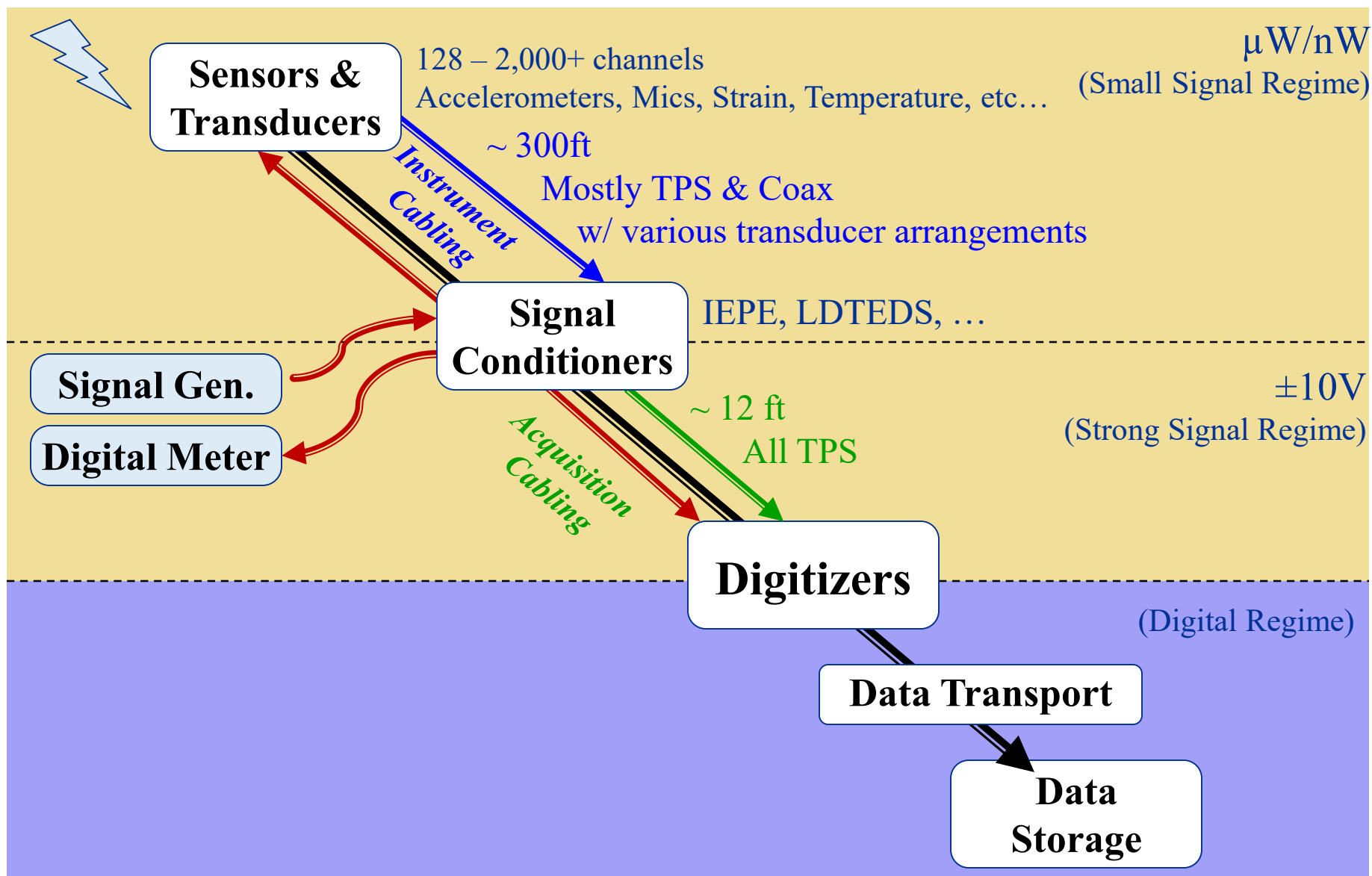




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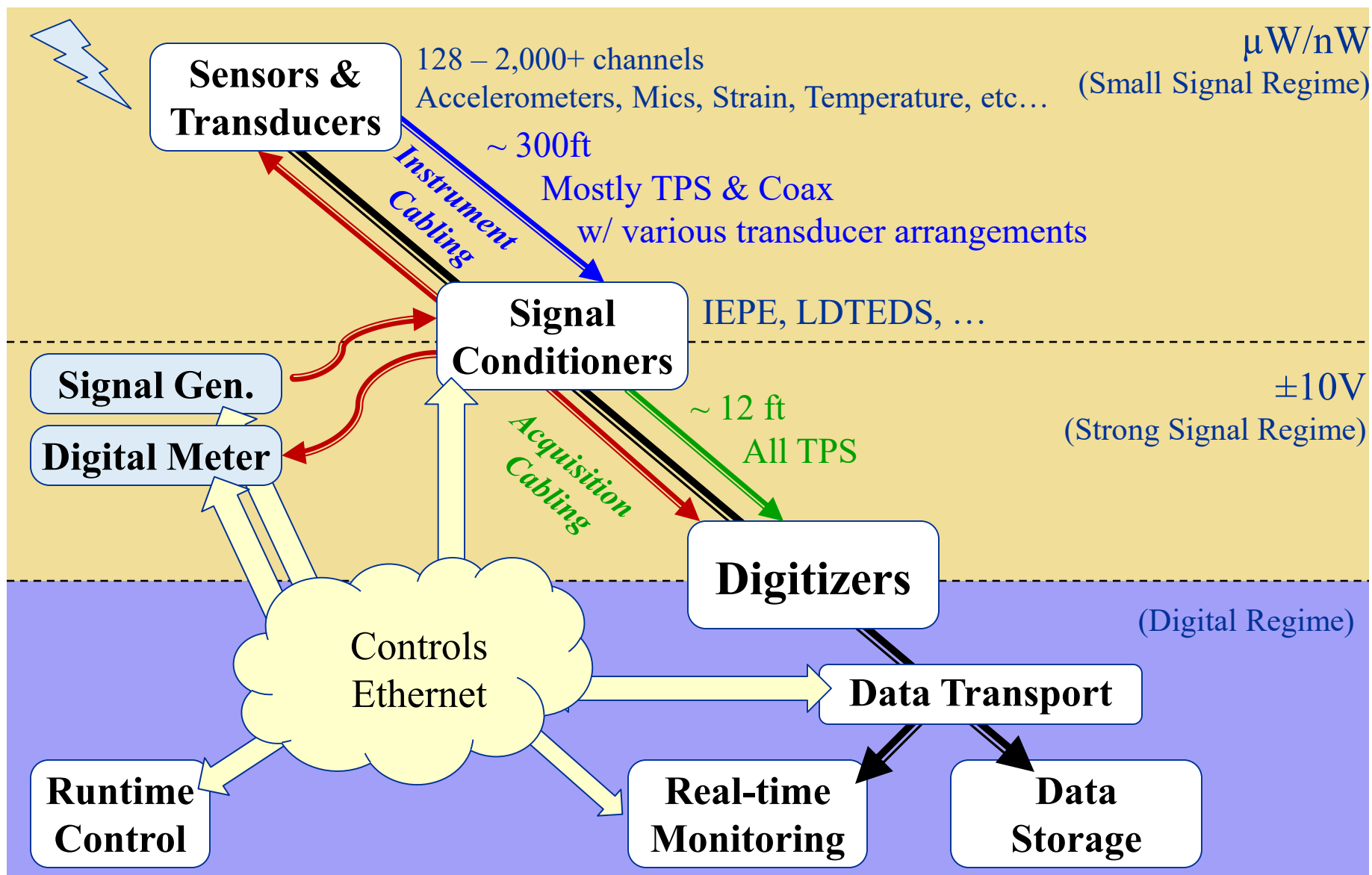


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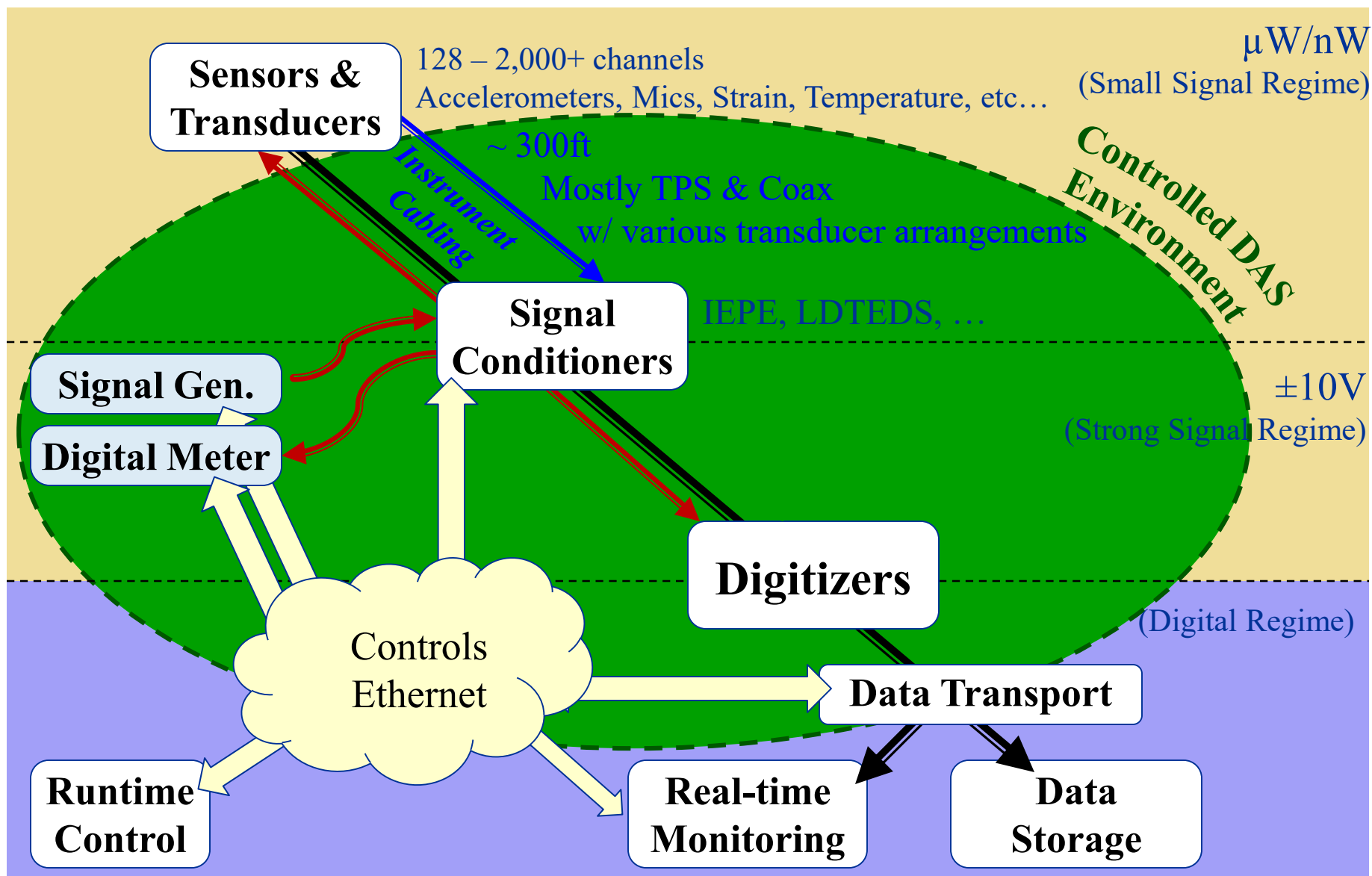




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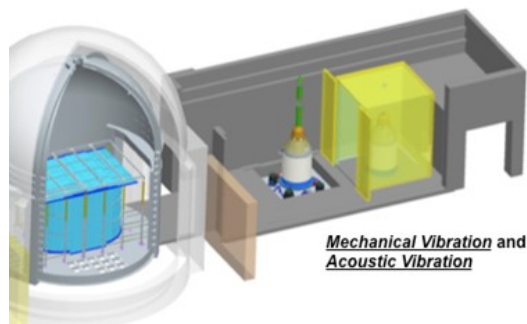


# “Wired” Measurement Topology





# “Wired” Measurement Topology (HSDAS)



28, 256, 512, 1,024 and ~ 2,000+ channels  
accelerometers, Mics, Strain, Temperature, etc...

$\mu\text{W/nW}$

(Small Signal Regime)

< 300ft

Mostly TPS & Coax

w/ various transducer arrangements

Controlled DAS Environment

Standardized Signal Conditioning/Ch.

16 bit ADCs

1,024 – 1,536 channels

variable SR/Ch.

$\pm 10\text{V}$

(Strong Signal Regime)

Acquisition Cabling

Controls  
Ethernet

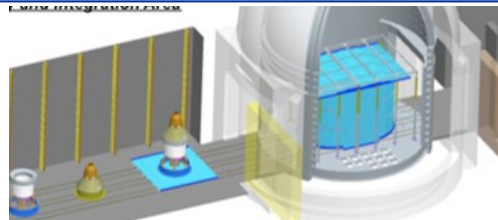
4Gbps/ea.

(Digital Regime)

3TB/ea.



# “Wired” Measurement Topology (LSDAS)



128, 256, 512, 1,024 and ~ 2,000+ channels  
Accelerometers, Mics, Strain, Temperature, etc...

$\mu\text{W/nW}$   
(Small Signal Regime)

< 300ft

Type “T” Thermocouples



(280)

Integrated Temperature  
Measurement System

ADAS Translator  
Brick/PC



Controlled DAS  
Environment

$\pm 10\text{V}$   
(Strong Signal Regime)

Ethernet  
Cabling

Controls  
Ethernet



4Gbps/ea. (Digital Regime)

3TB/ea.



# In-Space Propulsion (ISP) Test Facility - DAS



Designed for High Altitude Engine Testing



Instrument Cabling to Test Chamber		Quantities
1	Single Pair, Twisted-Pair Shielded	>1,700 Ch. 1PR TPS
2	Four-Wire, Twisted-Pair Shielded	312 Ch. of 4C / Ch.
3	Impedance Matched Coaxial	~64 Ch.
Signal Conditioning Equipment		
4	Constant-Voltage Bridge Conditioners	76 Ch.
5	Constant-Current Bridge Conditioners	32 Ch.
6	ICP/IEPE Conditioners	48 Ch.
7	Charge-Type Amplifiers	
8	Filter/Amplifier Signal Conditioners	160 Ch.
9	Frequency-to-Voltage	28 Ch.
10	Direct Voltage Inputs	600+ (see ADCs)
11	UTR Thermocouple Conditioners	<tb>
Digitizing Equipment (ADCs)		
12	High-Speed Digitizers (110 kHz MBW/Ch.)	32 Ch. - 256 Ch.
13	Low-Speed Digitizers (1 kHz MBW/Ch.)	576 Ch.
14	Discrete Channel Acquisition	32 Ch.
Data Storage		
15	RAID 1+0 redundant fail-over storage	2.5 – 3 Terabytes
Control, Monitoring and Post-Processing		
16	Dedicated Control Computers	2
17	Dedicated Monitoring Computers	4
18	Dedicated Post-Processing Computer	1
Other Notable System Elements		
19	IRIG-B Distribution	All
20	LTO-3 Tape Archival System	All

# Space Environments Complex (SEC) – HSDAS



Designed for Large-Scale Environment Testing



Instrument Cabling to the MVF table and RATF chambers		Num. of Channels
2	Impedance Matched Coaxial to/from ACS	64 Ch.
3	Impedance Matched Coaxial to/from VCS	64 Ch.
<b>Signal Conditioning Equipment</b>		
5	Constant-Voltage Bridge Conditioners	160 Ch.
6	Direct Voltage Inputs from ACS/VCS	64 Ch.
7	Buffered Voltage Outputs to ACS/VCS	64 Ch.
<b>Digitizing Equipment (ADCs)</b>		
<b>Data Storage</b>		
9	RAID 1+0 redundant fail-over storage	3 Terabytes total
<b>Control, Monitoring and Post-Processing</b>		
10	Dedicated Control Computers	1
11	Dedicated Monitoring Computers	3
12	Dedicated Post-Processing Computer	1
<b>Other Notable System Elements</b>		
13	IRIG-B Distribution	All
14	LTO-3 Tape Archive	All



# Space Environments Complex (SEC) – LSDAS

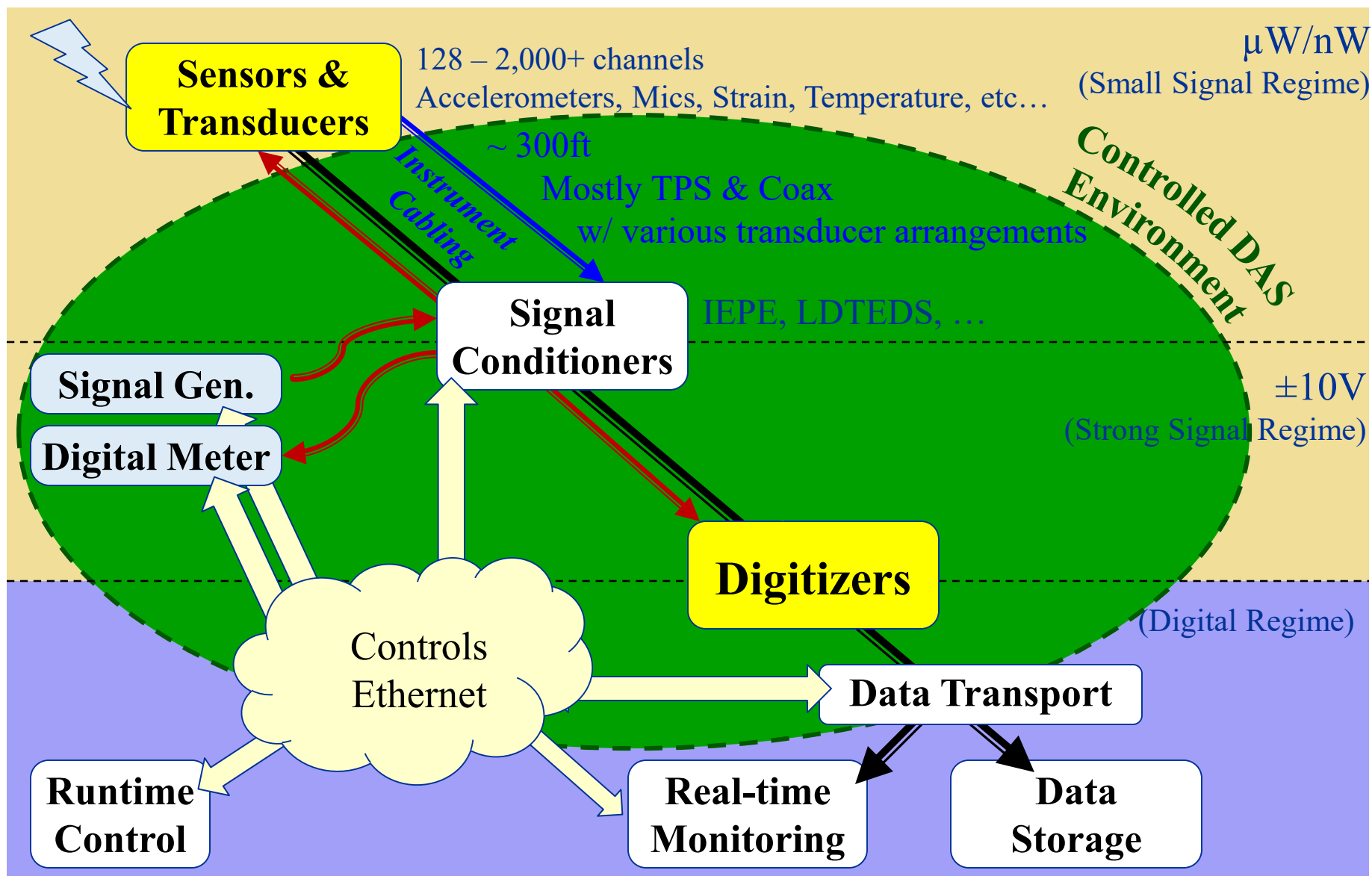


Designed for Large-Scale Environment Testing



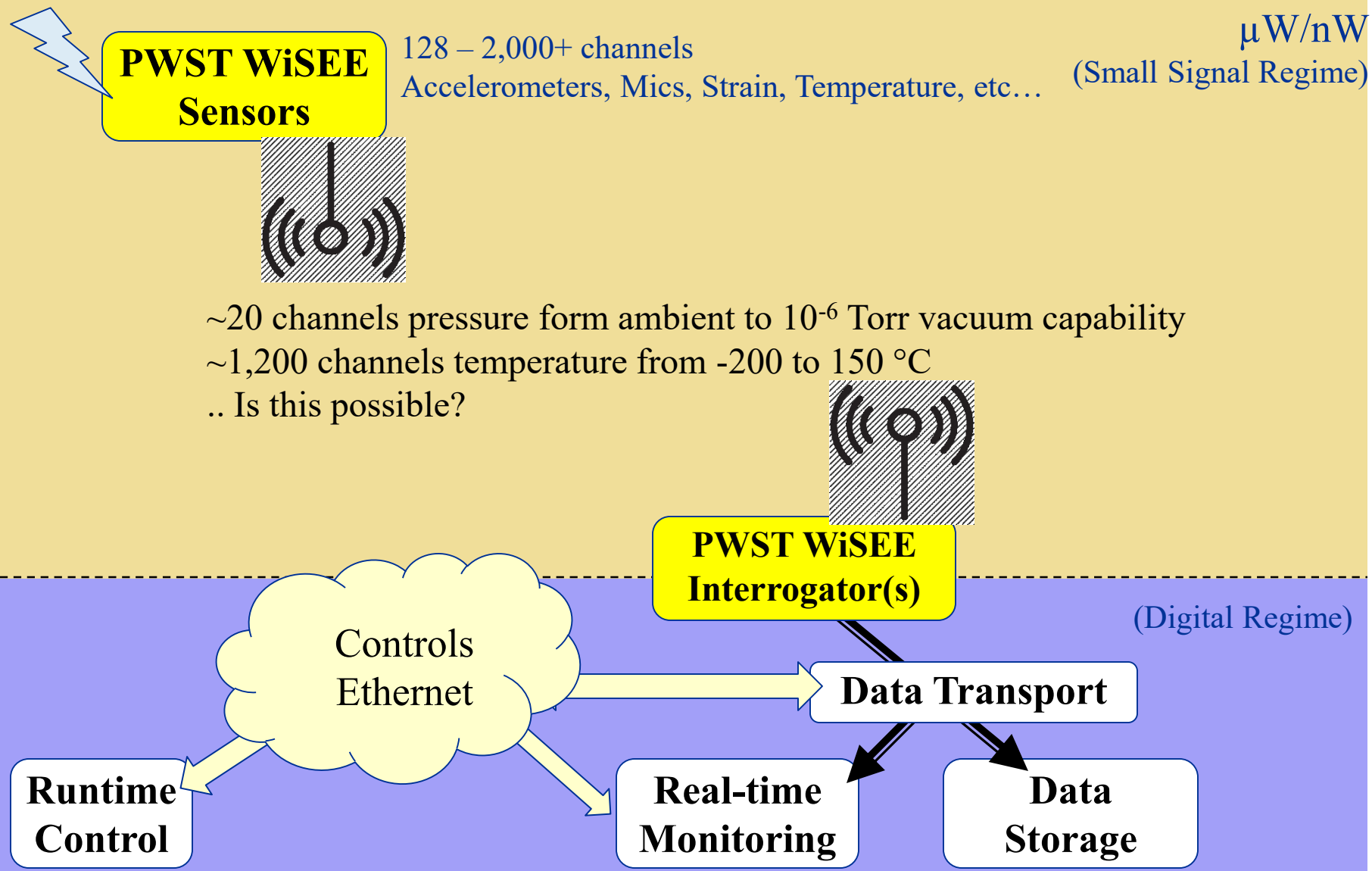
Instrument Cabling in the Vacuum Chamber Annulus		Num. of Channels
1	Single Pair, Twisted-Pair Shielded	324 Ch. 1PR TPS
2	Four-Wire, Twisted-Pair Shielded	288 Ch. 4C/Ch.
3	Impedance Matched Coaxial	126 Ch. BNC
		- 1,280 Ch.
<b>Signal Conditioning Equipment</b>		
5	ICP/IEPE Conditioners	64 Ch. *
6	Constant-Voltage Bridge Conditioners	48 Ch. †
7	Charge-Type Conditioners	24 Ch. †
8	Direct Voltage Inputs	120 Ch.
9	Buffered Voltage Follower Outputs	As needed †
10	Thermocouple Signal Conditioning	
<b>Digitizing Equipment (ADCs)</b>		
11a	Low-Speed Digitizers/Data-Translators (10Hz SR)	512 Ch. - 1,280 Ch.
<b>Data Storage</b>		
12	RAID 1+0 redundant fail-over storage	3 Terabytes total
<b>Control, Monitoring and Post-Processing</b>		
13	Dedicated Control Computers	1
14	Dedicated Monitoring Computers	4*
15	Dedicated Post-Processing Computer	1*
<b>Other Notable System Elements</b>		
16	IRIG-B Distribution	All
17	LTO-3 Tape Archive	All*

# Topology of “Wired” Measurements





# Topology of “Wireless” Measurements





*Thank you.*